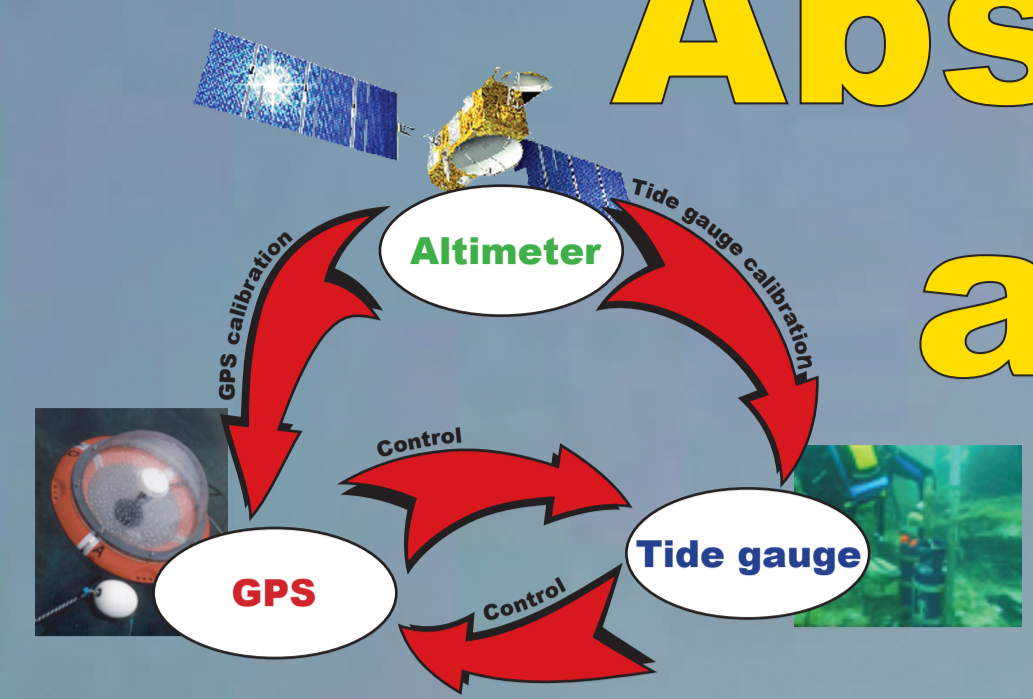
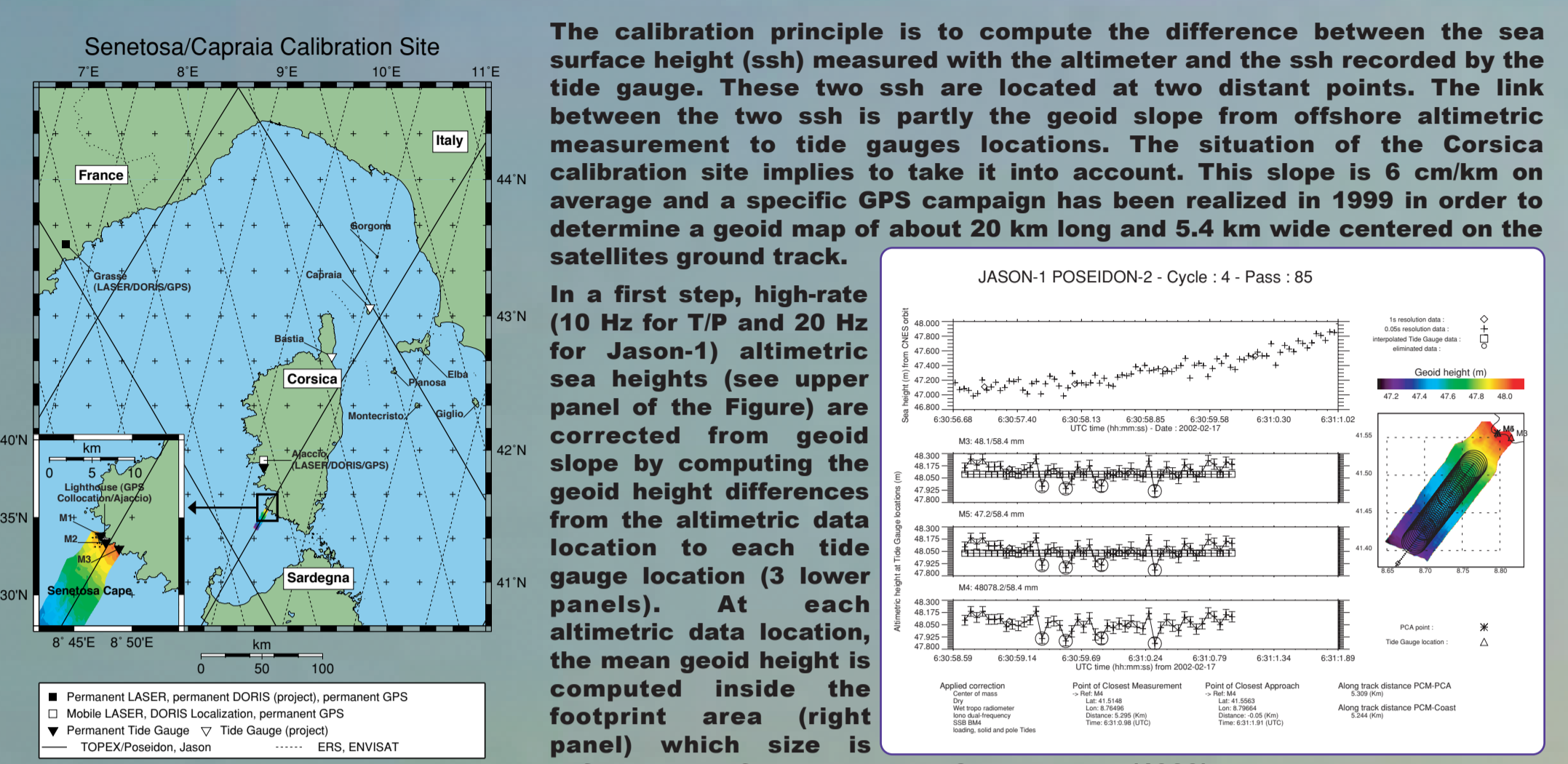


Absolute Calibration of Jason-1 and TOPEX/Poseidon Altimeters in Corsica



GENERAL OVERVIEW

Calibration process



The Senetosa calibration site provides in situ sea height estimations and local condition parameters (tide gauges, GPS buoy, meteorology station), that are compared to altimetric data.

Product	Notes	Definition
TOPEX/Poseidon	Use from NASA or CNES	Sea height in situ
Jason-1	Use from CNES (GDR/SSB)	Sea height in situ
IGDR	Use from CNES (IGDR/SSB)	Sea height in situ
SSALT	Use from NASA	Sea height in situ
GPS	Use from NASA	Sea height in situ
Tide gauge	Use from local tide gauges	Sea height in situ

The sea-height bias is thus defined by the following difference: altimeter sea height - in situ sea height. For example, a positive sea-height bias means that the altimetric sea height is erroneously high or the altimeter is measuring too short. Statistical computations for the sea-height bias have been realized using the following rules: (i) the bias for each overflight is obtained by averaging all the bias determinations from available tide gauge data, and (ii) the overall bias for each radar altimeter (POSEIDON-2, ALT-A, ALT-B and SSALT) corresponds to the mean (or median) of the estimates computed from all participating overflights (i.e., cycles). For all analyses, we have chosen to use data products provided by the T/P and Jason-1 projects. The principal assumptions are summarized below:

- Adopt Jason-1 GDR product as baseline
- Other orbits (GPS Reduced Dynamic from JPL and short-arc orbits from CERGA) have been used for comparisons only.

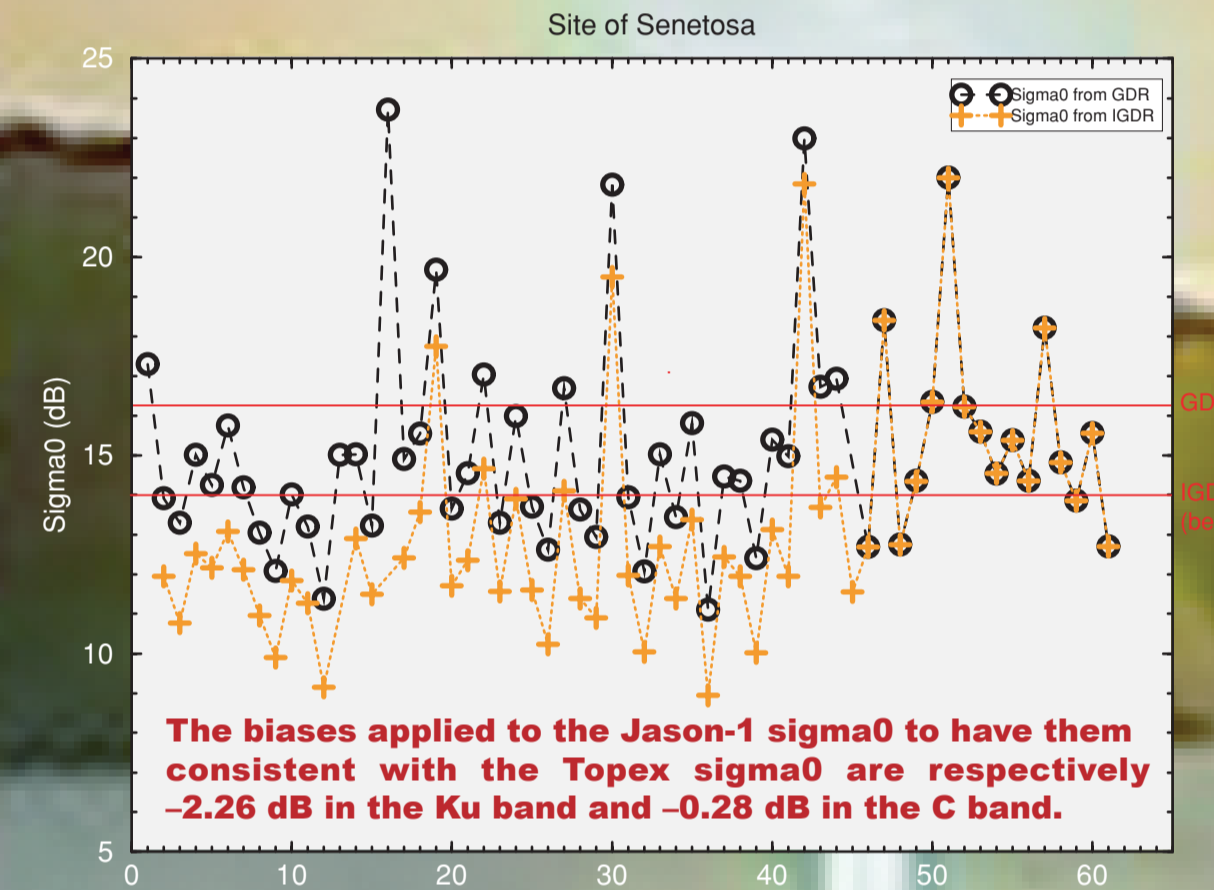
TOPEX/Poseidon:

- T/P M-GDR product is baseline. NASA precise orbit ephemeris has been chosen (Satellite Laser Ranging and DORIS data are used)
- Correct for drift in TOPEX Microwave Radiometer (TMR) data from 18-GHz channel instability (Ruf and Brown, 2002)

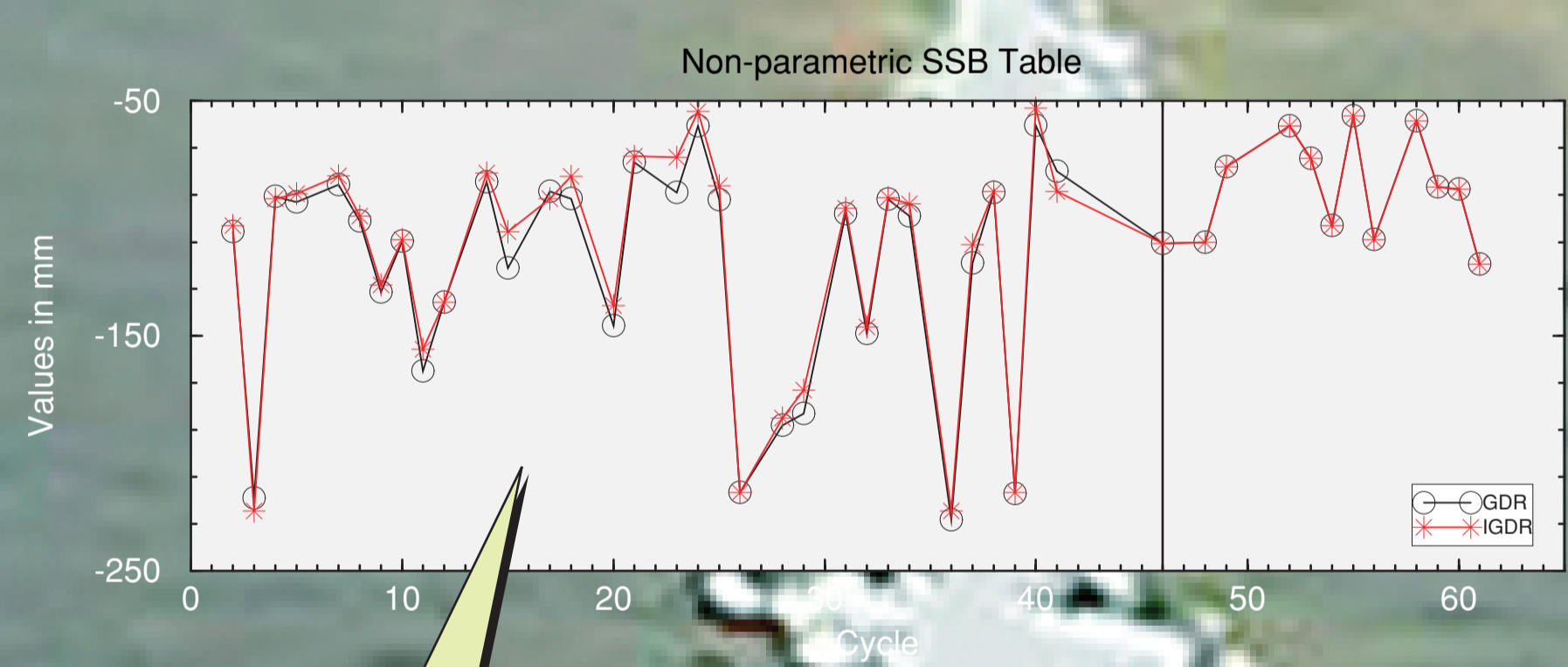
Finally, the altimeter bias time series are obtained after applying two-selection criterion: Only overflights with a sigma naught below 14 dB (16.26 dB for Jason-1) are kept avoiding a possible sigma bloom effect (see Figure below). This criteria leads to reject about 27% of the ALT (T/P), 36% of SSALT (T/P) and 14% of POSEIDON-2 (Jason-1) altimeters data.

To avoid erroneous tide gauges data, all cycles for which the standard error (issued from the averaging of tide gauges determinations) is higher than 10 mm are rejected (about 10%). This value has been chosen taking into account that the individual tide gauge measurement precision is better than 10 mm.

Jason-1 Sigma0



Sea State Bias



The SSB correction we used in replacement of the IGDR one comes from S. Labrousse study (CLS) and is implemented in IGDR production since cycle 46. The differences we observe before cycle 46 are due to differences in the SWH and wind speed which are used to compute SSB. The total contribution of these differences is +3 mm (bias computed from GDR is increased by 3 mm).

Impact of the Corrections

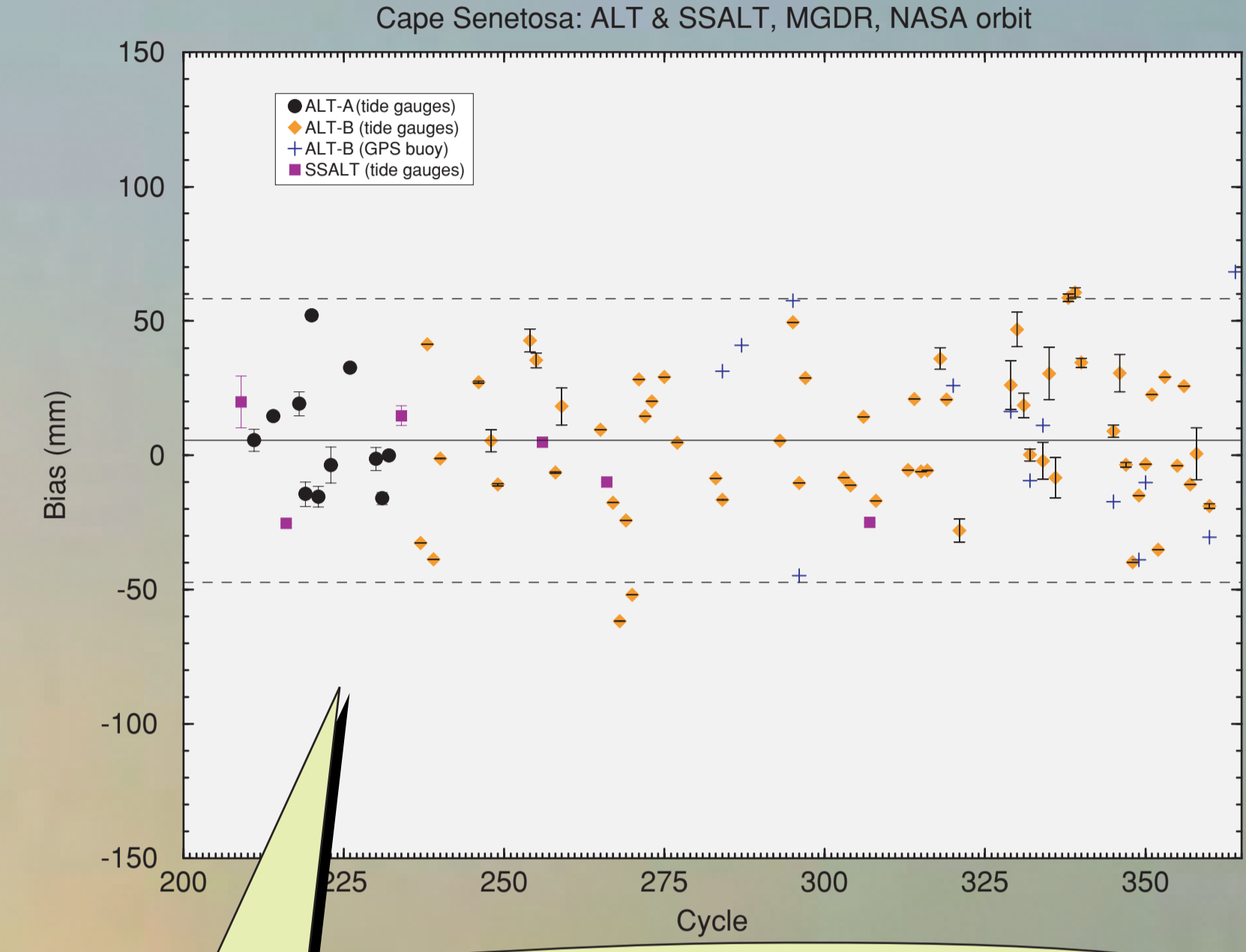
The total contribution of the differences between IGDR and GDR is negligible. However, the differences between T/P and Jason-1 corrections imply that POSEIDON-2 altimeter bias should be higher by 20 mm if T/P corrections are used in place of Jason-1 ones. The detailed contribution is:

- 2 mm from dry tropospheric path delay
- +13 mm from wet tropospheric path delay
- +8 mm from dry ionospheric path delay

The higher contribution is from JMR. Based on previous studies we have shown that the wet tropospheric path delay determined from GPS is very close to TMR (~4 mm). This suggests that even with the recent improvements the JMR is measuring shorter by about 10 mm, at least in this area.

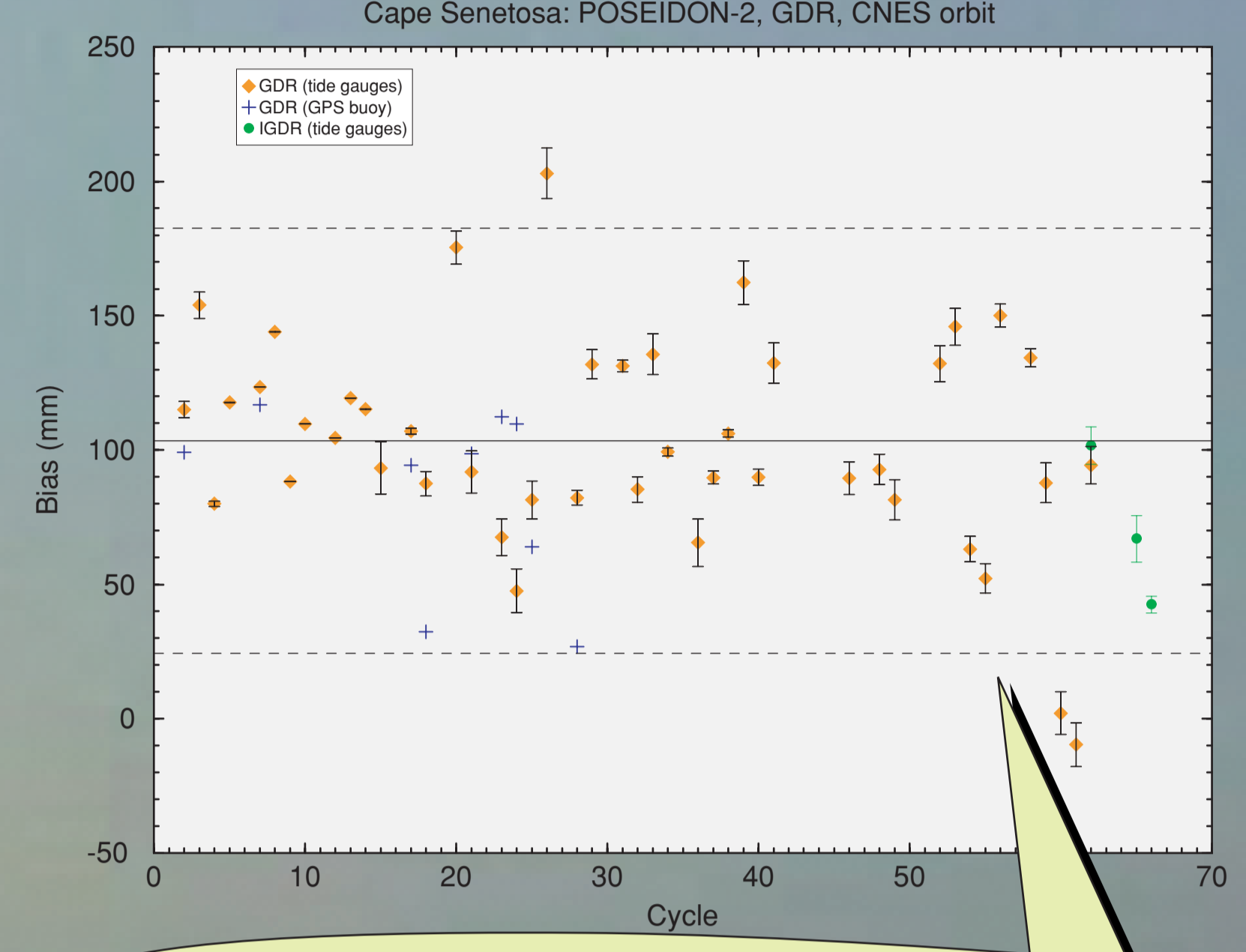
Altimeter Biases Time Series

TOPEX/Poseidon Altimeter Biases (M-GDR)



TOPEX/Poseidon altimeters (ALT-A and ALT-B) have been calibrated from 1998. Results show a great coherence between both altimeters +7 ± 7 mm, +6 ± 3 mm and -3 ± 8 mm for ALT-A, ALT-B and SSALT respectively. Moreover, results are very consistent with those obtained from the Harvest Platform (differences of few millimeters). For the 10 cycles in common, the tide gauges and GPS buoy determinations are respectively +2 ± 7 mm and -3 ± 10 mm. Results obtained by the two techniques thus show a very high consistency at the millimeter level, within the error bars.

Jason-1 Altimeter Biases (GDR)



POSEIDON-2 altimeter bias is +104 ± 6 mm, based on 62 cycles of GDR products. Consistency of both techniques appears to be lower than for T/P. However, this effect is only due to the time distribution of the GPS buoy: half the set is during relatively low values of Jason-1 altimeter bias (cycle 20 to 25). Indeed, when comparing the time series on common cycles, tide gauges determination only differ by few millimeters (5 mm higher). Thus, once again GPS buoy determination appears to be a very promising and powerful technique.

However, the standard deviation of the Jason-1 altimeter bias is higher than for TOPEX/Poseidon by about 15 mm (dotted lines on the above Figures). Indeed, a running average on the whole T/P bias time series with a window size of 62 cycles (Jason-1 sample size) even shows a very stable standard deviation of 25 mm. Considering that the ground measurement system is the same for both satellites, it means that the satellite measurements (range and corrections) are more noisy for Jason-1.

Impact of Orbit on Calibration

Orbit	Altitude (km)	SLR (km)	SLR (km)
1891	1330	1330	1330
2001	1330	1330	1330
2101	1330	1330	1330
2201	1330	1330	1330
2301	1330	1330	1330
2401	1330	1330	1330
2501	1330	1330	1330
2601	1330	1330	1330
2701	1330	1330	1330
2801	1330	1330	1330
2901	1330	1330	1330
3001	1330	1330	1330
3101	1330	1330	1330
3201	1330	1330	1330
3301	1330	1330	1330
3401	1330	1330	1330
3501	1330	1330	1330
3601	1330	1330	1330
3701	1330	1330	1330
3801	1330	1330	1330
3901	1330	1330	1330
4001	1330	1330	1330
4101	1330	1330	1330
4201	1330	1330	1330
4301	1330	1330	1330
4401	1330	1330	1330
4501	1330	1330	1330
4601	1330	1330	1330
4701	1330	1330	1330
4801	1330	1330	1330
4901	1330	1330	1330
5001	1330	1330	1330
5101	1330	1330	1330
5201	1330	1330	1330
5301	1330	1330	1330
5401	1330	1330	1330
5501	1330	1330	1330
5601	1330	1330	1330
5701	1330	1330	1330
5801	1330	1330	1330
5901	1330	1330	1330
6001	1330	1330	1330
6101	1330	1330	1330
6201	1330	1330	1330
6301	1330	1330	1330
6401	1330	1330	1330
6501	1330	1330	1330
6601	1330	1330	1330
6701	1330	1330	1330
6801	1330	1330	1330
6901	1330	1330	1330
7001	1330	1330	1330
7101	1330	1330	1330
7201	1330	1330	1330
7301	1330	1330	1330
7401	1330	1330	1330
7501	1330	1330	1330
7601	1330	1330	1330
7701	1330	1330	1330
7801	1330	1330	1330
7901	1330	1330	1330
8001	1330	1330	1330
8101	1330	1330	1330
8201	1330	1330	1330
8301	1330	1330	1330
8401	1330	1330	1330
8501	1330	1330	1330
8601	1330	1330	1330
8701	1330	1330	1330
8801	1330	1330	1330
8901	1330	1330	1330
9001	1330	1330	1330
9101	1330	1330	1330
9201	1330	1330	1330
9301	1330	1330	1330
9401	1330	1330	1330
9501	1330	1330	1330
9601	1330	1330	1330
9701	1330	1330	1330
9801	1330	1330	1330
9901	1330	1330	1330
10001	1330	1330	1330

Thanks to the FTLRS tracking support and the effort made by the European SLR network, short-arc orbits have been computed for 60% of the Senetosa overflights. The radial accuracy of short-arc orbits is strongly dependent to vertical station coordinate and range bias in the measurements: the total root sum square contribution of these parameters, for the FTLRS, has been estimated to be below 3 mm. Figure at right shows the Jason-1 altimeter bias for GPS Reduced Dynamic (GPS RD, JPL), Precise Orbit Ephemeris (POE, CNES), Short-Arc Orbits (SAO, CERGA) and Medium Orbit Ephemeris (MOE, CNES). The distribution of the biases from short-arc orbits is very coherent with POE and GPS RD. This figure also shows that the tracking support was more efficient during the Cal/Val phase, thanks to FTLRS, than for the following period (40% of the overflights).

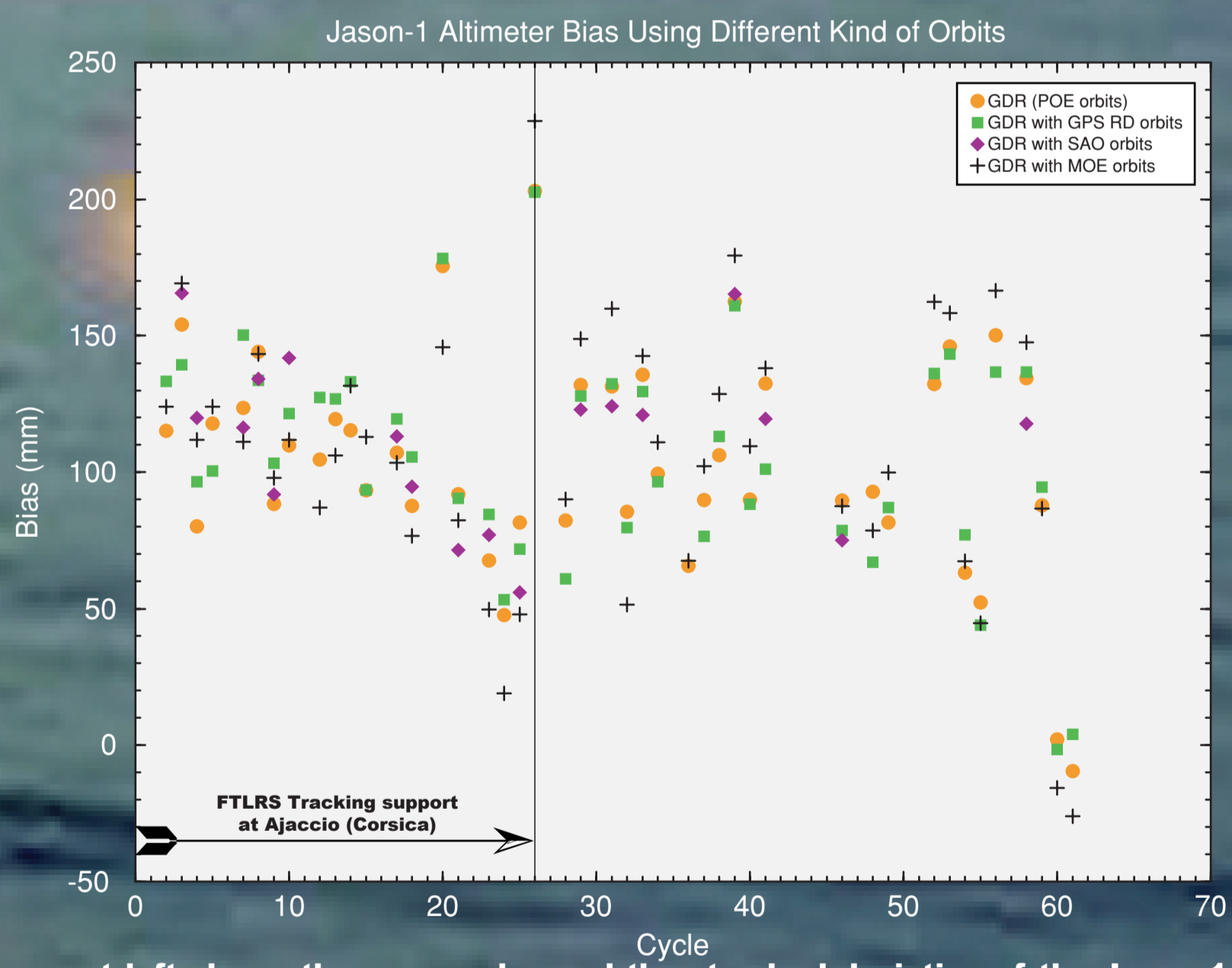


Figure at left shows the mean value and the standard deviation of the Jason-1 altimeter bias for each kind of orbits. All the results are within few millimeters and the relatively high value for short-arc orbits is due to the high density of determinations during the validation phase. For comparison, on cycle 1 to 26 mean bias from short-arcs and POE differs by only 4 mm. On the other hand, the short-arc solutions exhibit a standard deviation lower by 10 mm showing the high consistency and accuracy of the short-arc technique. More details on orbit impact, notably due to Geographically Correlated Errors are given in the poster:

Validation Activities for Jason-1 and TOPEX/Poseidon Precise Orbits

The whole calibration process (Tide gauges and GPS buoy) have been validated with TOPEX/Poseidon over 4 years of data and results are very consistent with Harvest ones. For Jason-1 all the GDR (62 cycles) have been analysed:

- Jason-1 (GDR): +7 ± 7 mm
- TOPEX/Poseidon (ALT-A): +6 ± 3 mm
- TOPEX/Poseidon (ALT-B): -3 ± 8 mm
- Jason-1 (POSEIDON-2): +104 ± 6 mm

Relative bias between Jason-1 and TOPEX/Poseidon missions is close to +100 mm. Results will be continuously updated through Jason-1 mission and can be consulted on the web site: <http://grasse.obs-azur.fr/cicerga/gmca/calval/alt/>

Analysis of Jason-1 Altimeter corrections

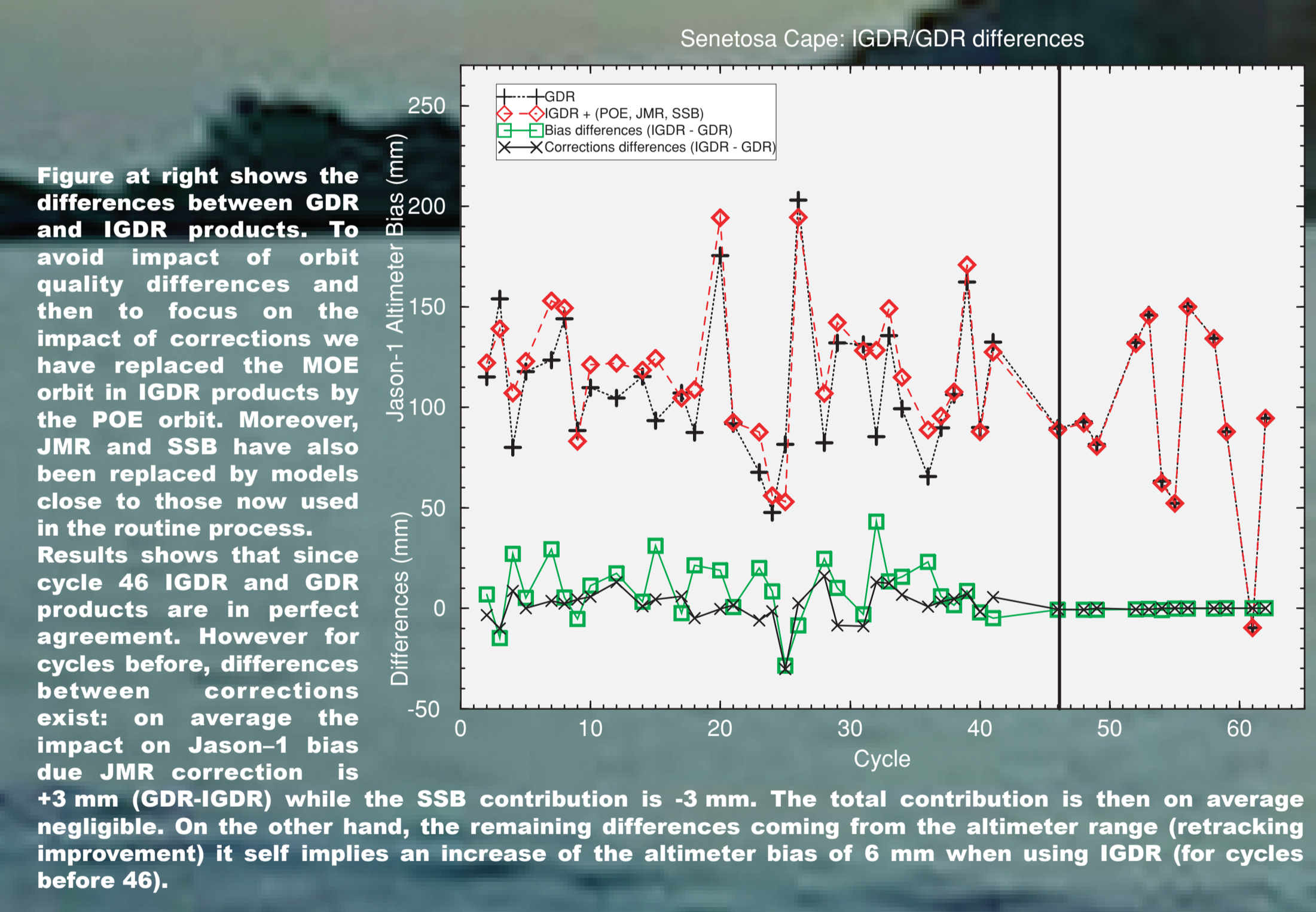
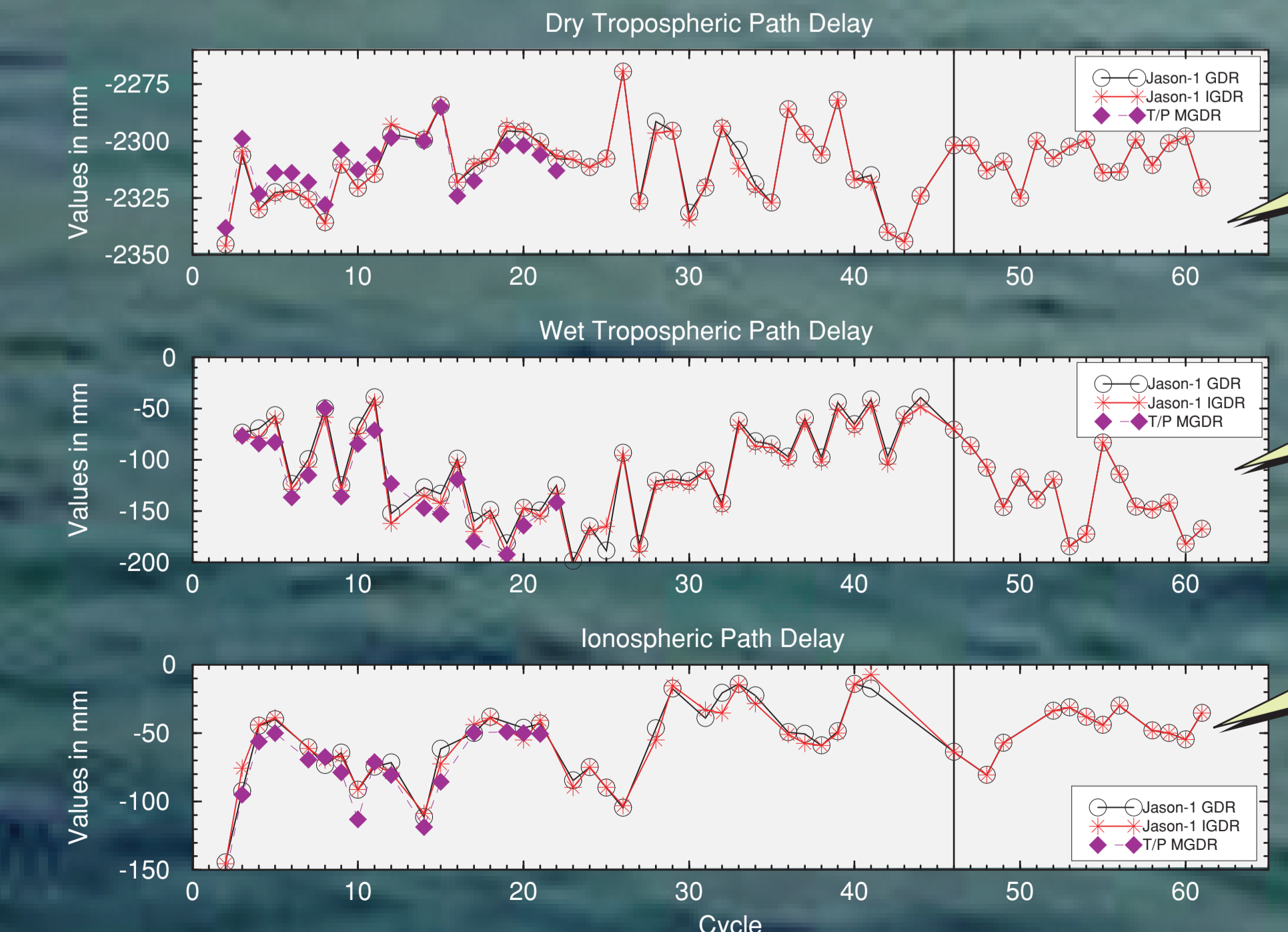


Figure at right shows the differences between GDR and IGDR products. To avoid impact of orbit quality differences and then to focus on the impact of corrections we have replaced the MOE orbit in IGDR products by the POE orbit. Moreover, JMR and SSB have also been replaced by models close to those now used in the routine process. Results shows that since cycle 46 IGDR and GDR products are in perfect agreement. However for cycles before, differences between corrections exist: on average the impact on Jason-1 bias due JMR correction is +3 mm (GDR-IGDR) while the SSB contribution is -3 mm. The total contribution is then on average negligible. On the other hand, the remaining differences coming from the altimeter range (retracing improvement) it self implies an increase of the altimeter bias of 6 mm when using IGDR (for cycles before 46).

Atmospheric Corrections



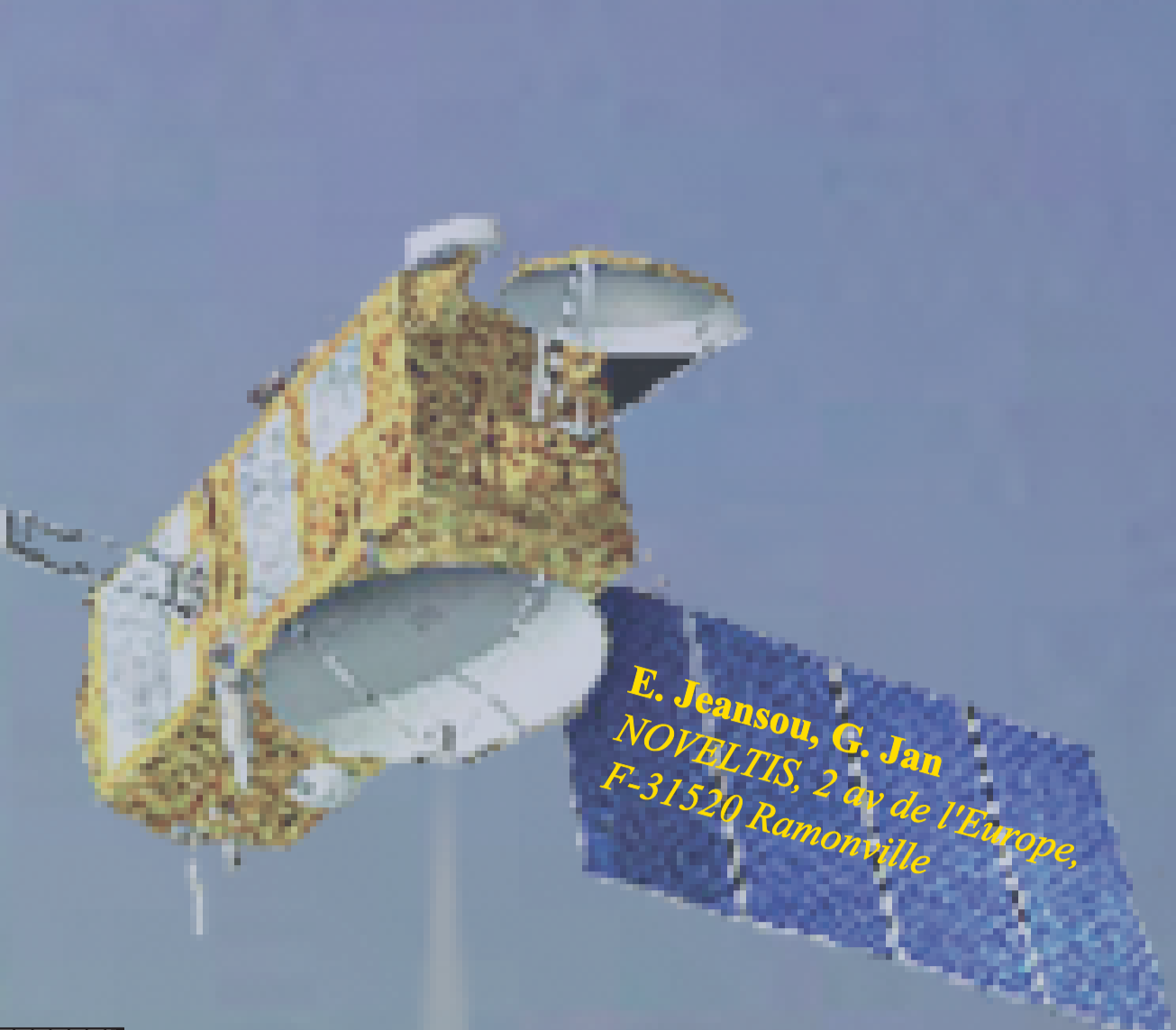
Averaged differences between GDR and IGDR is zero. When comparing with T/P during the tandem phase the path delay is higher by 2 mm for Jason-1.

Averaged differences between GDR and IGDR is -3mm. When comparing with T/P (TMR) during the tandem phase the path delay is lower by 13 mm for Jason-1 (JMR).

Averaged differences between GDR and IGDR is zero. When comparing with T/P during the tandem phase the path delay is lower by 8 mm for Jason-1.

Geophysical Corrections:

They are in perfect agreement for both IGDR and GDR products as well as for T/P (MGDR product)



Abstract

The double geodetic site in Corsica (Senetosa, near Ajaccio) has been used to calibrate the TOPEX/Poseidon (T/P) altimeters from 1998, and the Jason-1 ones since the beginning of the mission. Permanent and semi-permanent geodetic equipments are used to monitor these calibrations. Concerning the Aspreto site, a permanent GPS station and an automatic tide gauge have been installed since 1999. From January to September 2002, the French Transportable Laser Ranging System has been settled and its tracking support has permitted to locally improve orbit. A Senetosa cape, permanent geodetic installations have been installed since 1998 and different campaigns have been conducted in view of Jason-1 mission. Three tide gauges have been installed at the Senetosa Cape and linked to ITRF using GPS and leveling. In parallel, since 2000, a GPS buoy is deployed every 10 days at Senetosa (10 km off-shore). Besides, two GPS campaigns (1998 and 1999) have been performed to measure the marine geoid slope from the coast to 20 km off Senetosa cape - in this area the geoid slope can reach 6 cm/km. T/P altimeter calibration has been performed from cycle 208 to 365 using MGDR, and Jason-1 calibration has been performed from cycle 1 to 62 using GDR products. All the produced IGDR cycles have been also analysed with all the upgraded corrections used for GDR production (SSB, JMR, POE orbits). Our semi-permanent experiment is planned to last over several years in order to detect any drift in the space borne instruments.