

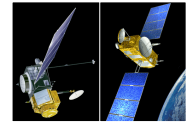


Improvement of the Complete TOPEX/POSEIDON and Jason-1 Orbit Time Series: Current Status

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ABSTRACT

Orbit error is a major component in the overall error budget of all altimeter satellite missions. Jason-1 is no exception and a 1 cm radial orbit accuracy goal has been set, which represents a significant improvement over what is currently being achieved for TOPEX/Poseidon (TP). Studies have demonstrated this goal is being met and that the orbit accuracies can be improved (Luthcke et al. 2003 and Haines et al. 2004). However, the challenge is to continually achieve this high accuracy, verify the performance, and characterize and quantify the remaining errors over the lifetime of the mission. The computation, verification and error characterization of such high accuracy orbits requires the reduction and analysis of all available tracking data (GPS, SLR, DORIS and altimeter). Current analysis also indicates the history of TP orbits can be further improved employing new solution strategies developed and tested on Jason-1. Our research focuses on the calibration, validation and improvement of the complete TP and Jason-1 orbit time series using all available tracking data including altimetry. Our effort will result in a complete and consistent time series of improved orbits for both TP and Jason, significantly benefiting the long time series of altimetry data records. The resultant high accuracy orbits and the characterization of their error will allow further improvements to the accuracy and overall quality of the altimeter measurement time series making possible further studies in radar altimeter remote sensing. In this presentation we summarize the current status of our research effort.

Improvement of the Complete TP Orbit Time Series : Results from a recent reprocessing

Overview: Table 1 presents the current modeling upgrades used to compute our latest complete TP orbit time series (cycles 1 through 446) based on a dynamic solution reduction of SLR+DORIS data. While several additional improvements are planned, the following Tables and Figures demonstrate the new TP orbits represent considerable improvement over the TP GDR orbits.

Models	GDR orbits	TVG New 2006
Gravity (static)	JGM3 (70x70)	GGM02C (120x120) Tapley et al. 2004
Gravity (time-variable)	C20dot, C21dot, S21dot	C20dot, C21dot, S21dot + 20x20 annual terms from GRACE inter-sat. tracking only (Tapley et al. 2005)
Atmospheric gravity	Not applied	NCEP S04S0 QRS files (Pavoni and Boy, 2004)
Ocean Tides	Ray 94 + GEMT3X	(TP-derived) GOT00.2 (20x20) (Ray and Poincu, 2002)
Solid Earth tides	$k_1=0.300$, $k_2=0.093$ + special handling for FCN	IERS2003
Station Coordinates	CSR98L02 (2001-2000) ITRF2000	ITRF2000

Table 2. TP Tracking Data Summary Statistics Cycles 001-446

	DORIS		SLR	
	Points	Residual (mm/s)	Points	Residual (cm)
GSFC SLR/DORIS dynamic Orbits				
gdr				
TP Standards (JGM3, Ray 94 tides, CSR98 stations to cycle 360, ITRF2000 stations from cycle 360)	24952008	0.5246	2065219	2.218
ggm02c	25695843	0.5010	2391924	1.963
ggm02c+ (GGM02C, ITRF2000 stations, Earth & Ocean (GOT00.2) tides conform to IERS2003)				
tv	25698634	0.5012	2392355	1.880
tv (As ggm02c+, plus IAU2000 reference, forward gravity modeling of the atmosphere, GSFC GRACE-derived annual time varying gravity)				

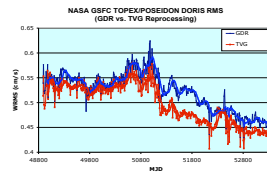
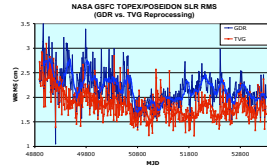
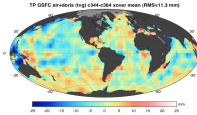
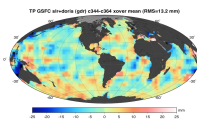


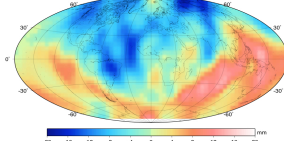
Table 2b. TP Tracking Data Summary Statistics Cycles 344-364

SLR/DORIS orbits	doris (mm/s)	slr (cm)	crossover (cm)
GDR (dynamic)	.4819	2.420	5.618
tv (dynamic)	.4799	1.816	5.442

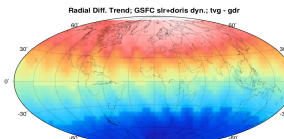


Impact on long-term sea level studies: The impact of the new orbits on sea level studies can, in large part, be characterized through the analysis of orbit differences between the new 'TVG' and the old 'GDR' orbits. The figures below present mean, trend and annual amplitude observed from the cycle by cycle gridded mean radial orbit differences computed from cycle 1 through 446.

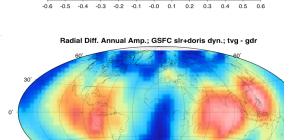
Mean Radial Diff: GSFC slr+doris dyn., tvg - gdr



Radial Diff. Trend: GSFC slr+doris dyn., tvg - gdr



Radial Diff. Annual Amp: GSFC slr+doris dyn., tvg - gdr



Improvement of the Jason-1 Orbit Time Series

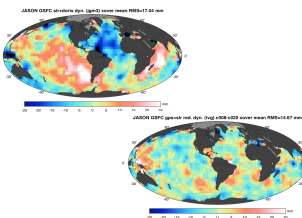
Overview: For Jason-1 the modeling upgrades described in Table 1 were implemented to update the POD solution models and strategy outlined in Luthcke et al. 2003. We have computed a new Jason-1 orbit time series (cycles 1 through 135) based on a dynamic solution reduction of SLR+DORIS data. We have also computed SLR+DORIS reduced dynamic, GPS-only reduced dynamic and GPS+SLR reduced dynamic orbits within the TP and Jason inter-comparison period (cycles 1-21). Again, while several additional improvements are planned, the following Tables and Figures demonstrate the new Jason-1 orbits represent an improvement over previous orbit generations. The results indicate the GPS-based orbits are exceeding the 1-cm radial orbit accuracy goal and improvement in SLR+DORIS based solutions have been made.

Table 3. Jason-1 Tracking Data Summary Statistics Cycles 001-135

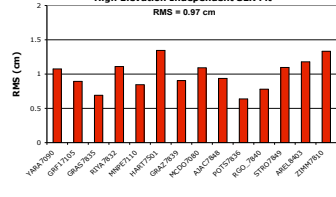
	DORIS		SLR	
	Points	Residual (mm/s)	Points	Residual (cm)
GSFC SLR/DORIS dynamic Orbits				
ggm01s	13387375	3990	506557	1.634
ggm01s (GGM01S, GOT99.2 tides, ITRF2000 stations)				
ggm02c	13423135	4151	506878	1.546
ggm02c+ (GGM02C, ITRF2000 stations, Earth & Ocean (GOT00.2) tides conform to IERS2003)				
tv	13393169	4071	506708	1.481
tv (As ggm02c+, plus IAU2000 reference, forward gravity modeling of the atmosphere, GSFC GRACE-derived annual time varying gravity)				

Table 4. Jason-1 Tracking Data Summary Statistics Cycles 008-020

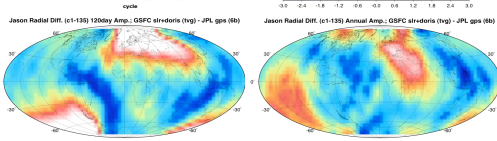
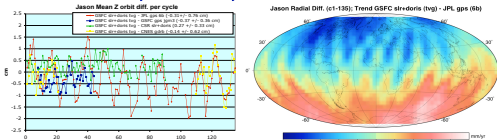
Solution	doris (mm/s)	slr (cm)	crossover (cm)
GSFC slr/doris ggm01s (dynamic)	4124	1.422	5.864
GSFC slr/doris tv (dynamic)	4120	1.331	5.807
GSFC slr/doris tv (reduced-dynamic)	4109	1.413	5.784
CNES GDR-B gps/slr/doris (dynamic)	4126	1.579	5.765
GSFC gps/slr tv (reduced-dynamic)	4123	1.108	5.712
JPL gps release_6b	4131	1.403	5.719
GSFC gps rd3	4123	1.300	5.717



GSFC GPS RD3 (tv) High Elevation Independent SLR Fit RMS = 0.97 cm



Orbit uncertainty and impact on sea level studies: Tables 3 and 4, and the Figures presented, demonstrate significant improvement in orbit accuracy and good agreement between various solution strategies and orbits computed from different centers. However, uncertainties in the orbits still remain that are troublesome for sea level studies. This is demonstrated below through the analysis of long-term (cycles 1 through 135) orbit differences between the most recent generations of the JPL GPS-only reduced dynamic orbits and the NASA GSFC SLR+DORIS dynamic orbits. It is important to note that the comparison does not target a problem solution, but characterizes the remaining orbit uncertainty. Problems and errors could equally exist in the NASA GSFC and JPL orbits. Future analysis, as well as model and solution strategy improvements will be made in order to further reduce the orbit uncertainties. The success in large part will depend on the continued diligence and cooperation of the OSTM POD Team members: CNES, NASA GSFC, JPL, UT CSR...



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Future: Future analysis, as well as model and solution strategy improvements will be made in order to further reduce the orbit uncertainties. The success, in large part, will depend on the continued diligence and cooperation of the OSTM POD Team members: CNES, NASA GSFC, JPL, UT CSR...

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